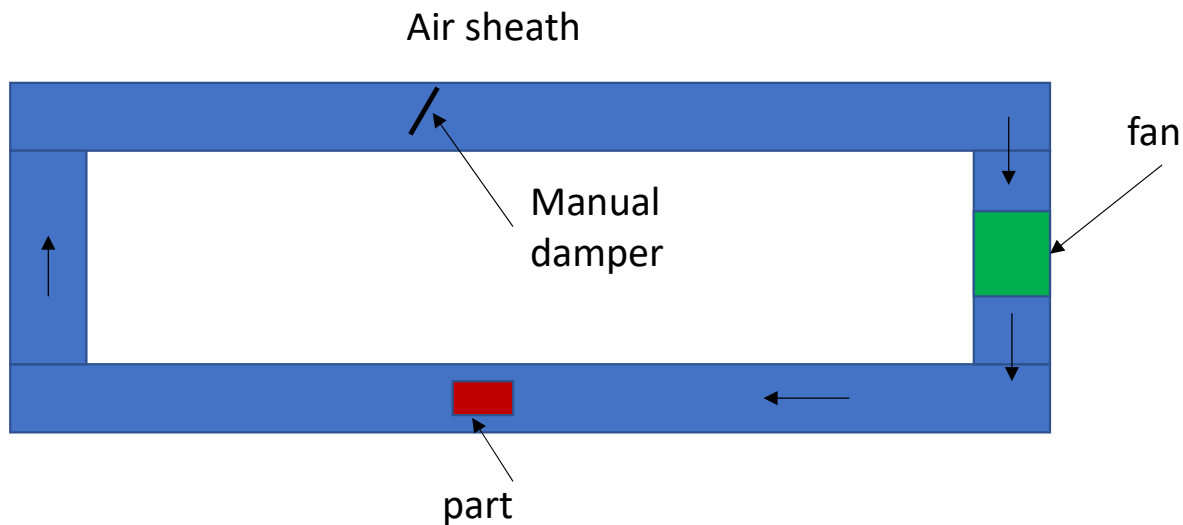


Heating tests of simple geometry parts at 150°C in forced convection in an air sheath



- The air speed in the air stream is controlled by a fan with frequency inverter + manual dampers
- The speed is measured by Pitot probes averaging along the horizontal and vertical axes.
- The air sheath has a section of 400 x 400 mm. The air sheath is mechanically welded in sheets steel, without external insulation. -> The wall temperature of the sheath is measured close to the parts to take into account the average radiation temperature.
- The temperature of the part is measured at several points (a point in the mass is taken as a reference), the difference between all the measurement points is negligible; the conduction in the volume is greater than the exchanges by convection. Number of Biot < 0.1



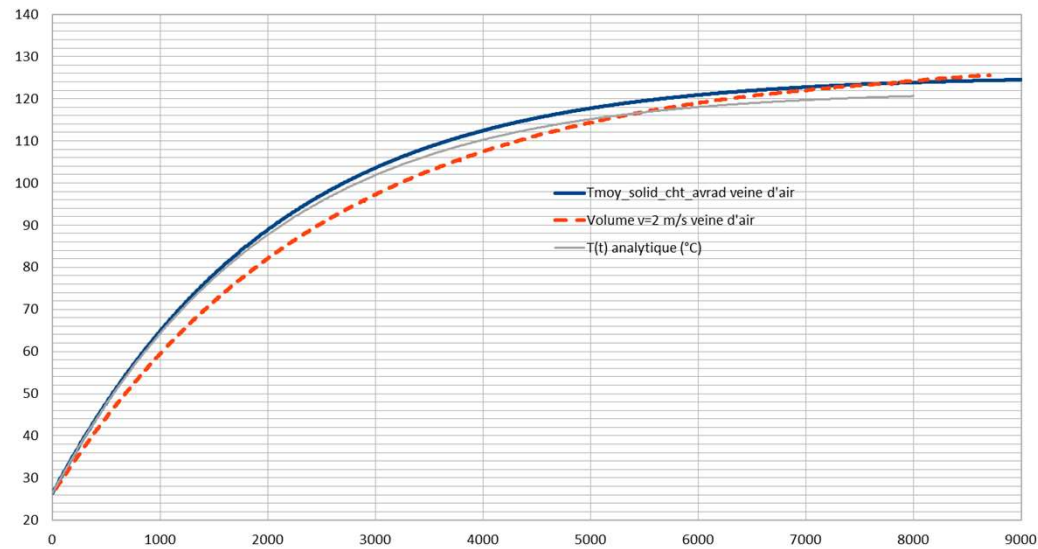
- We study simple shapes to be able to use the known correlations of the Nusselts number and to cross-check the simulations with the theory.

$$\overline{Nu} = 0.927 Re^{1/2}$$

- Case of a cylinder in an axial air flow
- The free software OpenFoam is used for comparison with the experimental data _ solver CHTSimpleFoam.
- The mesh is carried out with SnappyHexMesh and the CAD with Salome (EDF).
- The flow is turbulent, the chosen model is komegaSST with wall law. ($y^+ = 30$) Re varies between 10,000 and 40,000
- Natural convection is neglected
- The flow is calculated stationary by fixing the thermal properties of the air at the heating temperature. (density, viscosity and constant thermal conductivity) -> Frozen flow
- When the flow is stabilized, we use the transient solver to calculate the heating of the part
- The radiation are taken into account by a heat source on the surface of the part (radiation with the ambient we are in the situation of a closed box, no need of surface to surface radiations)

- Air velocity impact

$V_{\text{air}} = 2 \text{ m/s}$



Conditions:

$T_{\text{air}} = 139.5^{\circ}\text{C}$

$T_{\text{ray}} = 81.3^{\circ}\text{C}$

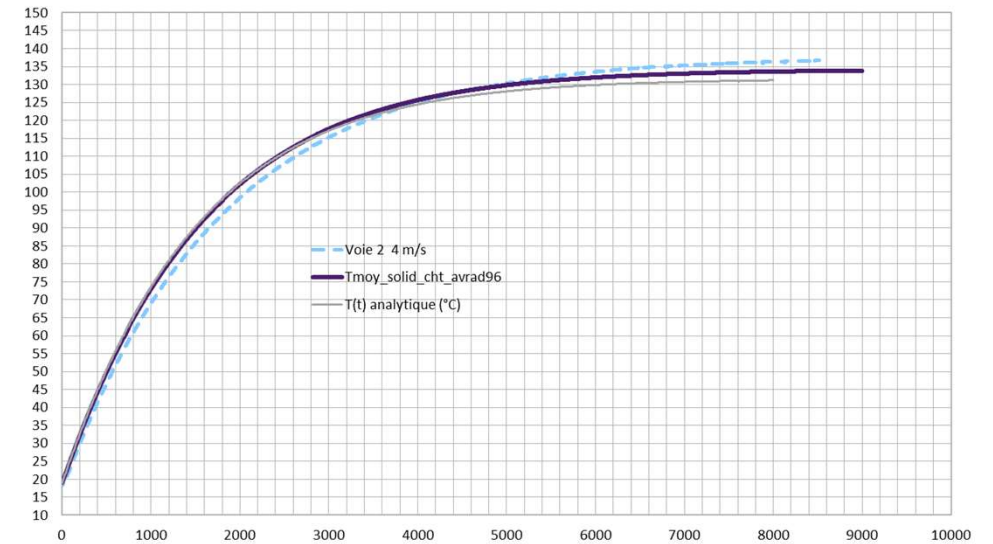
$T_{\text{ini}} = 26.5^{\circ}\text{C}$

$\varnothing 120 \text{ mm}$ $L_g 110 \text{ mm}$

Steel

$\text{Emi} = 0.9$

$V_{\text{air}} = 4 \text{ m/s}$



Conditions:

$T_{\text{air}} = 143.5^{\circ}\text{C}$

$T_{\text{ray}} = 96^{\circ}\text{C}$

$T_{\text{ini}} = 19^{\circ}\text{C}$

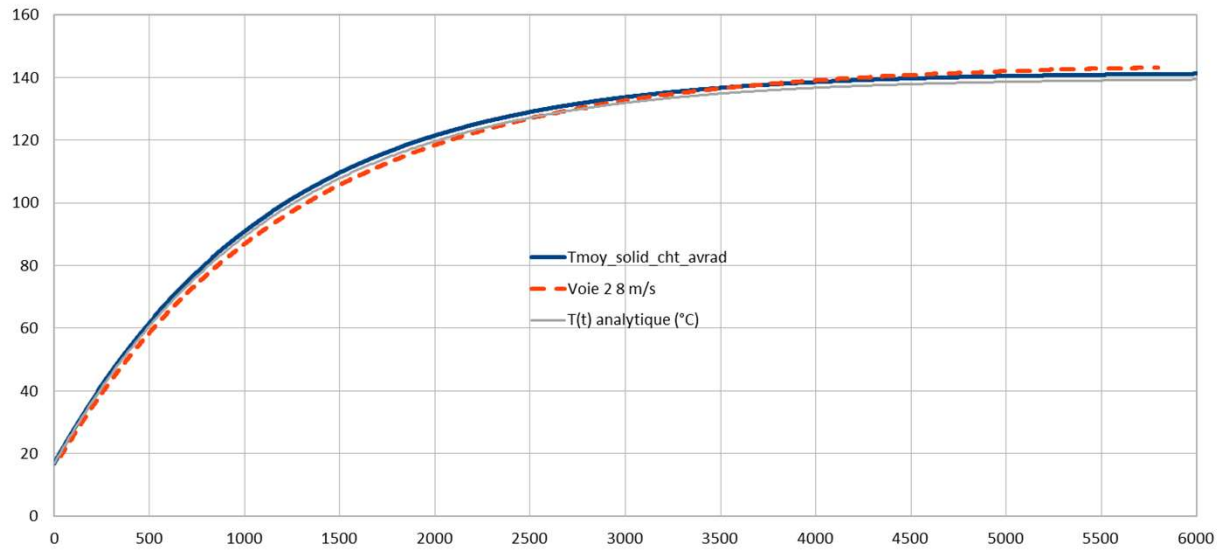
$\varnothing 120 \text{ mm}$ $L_g 110 \text{ mm}$

Steel

$\text{Emi} = 0.9$

- Air velocity impact

$V_{\text{air}} = 8 \text{ m/s}$



Conditions:

$T_{\text{air}} = 147.5^{\circ}\text{C}$

$T_{\text{ray}} = 110^{\circ}\text{C}$

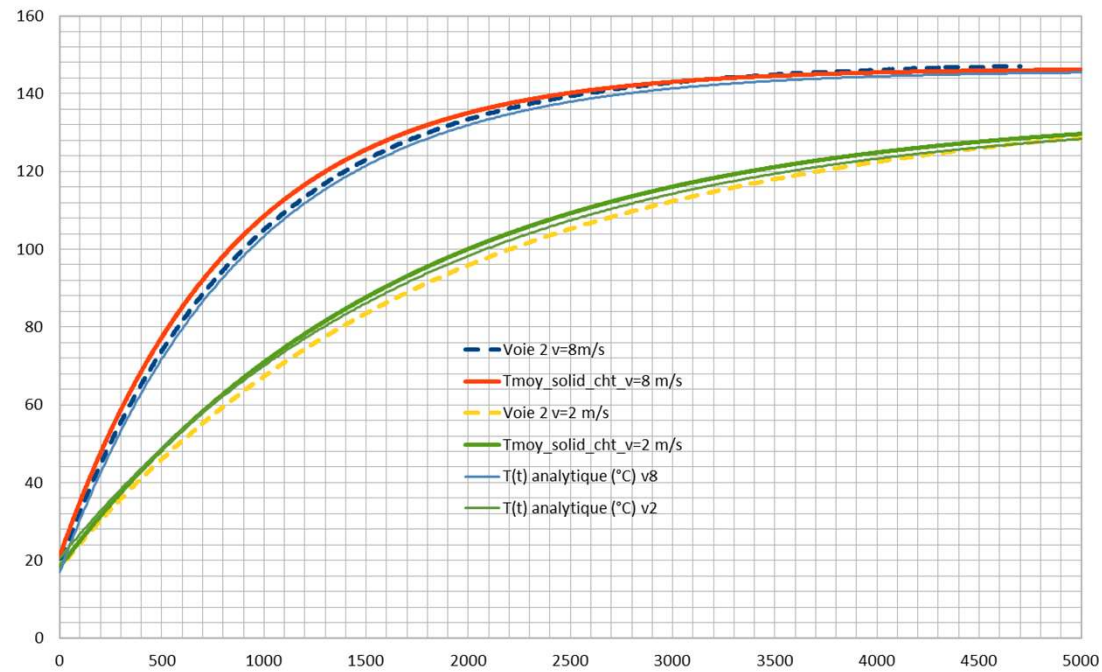
$T_{\text{ini}} = 17^{\circ}\text{C}$

$\varnothing 120 \text{ mm}$ $L_g 110 \text{ mm}$

Steel

$E_{\text{mi}} = 0.9$

- Material impact



Conditions:

$V=2 \text{ m/s}$

$T_{air} = 138.8^\circ\text{C}$

$T_{ray} = 82.2^\circ\text{C}$

$T_{ini} = 20.5^\circ\text{C}$

$\varnothing 120 \text{ mm}$ $L_g 110 \text{ mm}$

ALUMINIUM

$Emi = 0.15$

Conditions:

$V=8 \text{ m/s}$

$T_{air} = 147.5^\circ\text{C}$

$T_{ray} = 113^\circ\text{C}$

$T_{ini} = 21^\circ\text{C}$

$\varnothing 120 \text{ mm}$ $L_g 110 \text{ mm}$

ALUMINIUM

$Emi = 0.15$