EDF R&D



Fluid Dynamics, Power Generation and Environment Department Single Phase Thermal-Hydraulics Group

6, quai Watier F-78401 Chatou Cedex

SEPTEMBER 2017

Code_Saturne documentation

Code_Saturne version 5.0 tutorial: Shear Driven Cavity Flow

contact: saturne-support@edf.fr



EDF	R&D
$\mathbf{D}\mathbf{D}\mathbf{r}$	Iter

TABLE OF CONTENTS

	I Introduction	3
1	Introduction	4
1.1	TUTORIAL COMPONENTS	4
1.2	TUTORIAL STRUCTURE	4

	II Setting up	5
1	Setting up	6
1.1	What you will learn	6
1.2	USING SALOME _CFD TOGETHER FOR END-TO-END SIMULATIONS	6
1.3	INSTALLING SALOME _CFD OR Code_Saturne and SALOME SEPARATELY	6
1.4	GETTING STARTED WITH END-TO-END SALOME/ Code_Saturne CFD SIMULATIONS	7

	III Shear Driven Cavity Flow Case with CFDStudy	11
1	General description	12
1.1	What You Will Learn	12
1.2	CASE DESCRIPTION	12
1.3	CREATING THE Code_Saturne CASE	13
2	Creating the Computational Domain within SALOME	13
2.1	Geometry	14
2.2	Meshing	20
3	Simulating within SALOME	24
3.1	SETTING UP THE CFD SIMULATION	24
3.2	Running and Analysing the Simulation	34
3.3	Post-processing the Results	35
4	References	47
5	Reference Data from [4]	47

Part I

Introduction

1 Introduction

1.1 Tutorial Components

This tutorial makes use of:

- The SALOME [1] platform for geometry generation, meshing, and post-processing
- Code_Saturne [2], [3] for CFD calculations
- Reference [4] for comparison with published results

To work through this tutorial you will need a computer on which these two software applications are already available or on which you have permission to install them.

1.2 Tutorial Structure

This tutorial is made of two complementary sections:

- Section II describes all the procedures required to get going with setting up CFD simulations using SALOME and *Code_Saturne*, from code download to case creation.
- Section III illustrates setting up, running, and analysing a CFD simulation entirely with SA-LOME and *Code_Saturne*, using the laminar, Shear Driven Cavity as an example. This case is simple but contains enough physics to make it interesting and relevant to practical problems.

If you are already familiar with setting up CFD simulations with *Code_Saturne* and SALOME, you may skip Section II and go directly to Section III.

Part II

Setting up

1 Setting up

The first part of the tutorial is designed to explain the preliminary steps to setup an end-to-end CFD simulation using *Code_Saturne* and SALOME and provide an introduction to these software applications.

1.1 What you will learn

In the first part of the tutorial, you will learn:

- How *Code_Saturne* and SALOME may be combined in SALOME _CFD for end-to-end CFD analyses, from CAD generation to post-processing of CFD results
- How to access and download *Code_Saturne* and SALOME
- How to install SALOME and *Code_Saturne* within the CFDStudy module of SALOME
- How to set up a *Code_Saturne* case with CFDStudy

1.2 Using SALOME _CFD together for End-to-End Simulations

Code_Saturne is a general, 3D CFD solver which can be applied to a large variety of problems involving fluid flow and heat transfer. The *Code_Saturne* Graphical User Interface (GUI) can be used on its own to define CFD cases, and to run them when a mesh is already available. It has also been integrated as a module called CFDStudy in the SALOME open-source host platform, where it links with other modules to provide a single, user-friendly, graphical environment for end-to-end CFD analyses:

• **GEOM:** CAD module

 \circ Geometry and computational domain definition

• **SMESH:** Meshing module

 \circ Surface and volume meshing of the computational domain and defining boundaries

• **CFDStudy:** CFD module

 \circ Set up and run $Code_Saturne$ CFD calculations

- Paravis: Post-processing module
 - \circ Visualisation and analysis with Paraview [5]

The use of the different modules to set-up and run *Code_Saturne* studies is described throughout the different tutorials, starting with how to initiate a study with CFDStudy, as explained in 1.4.

1.3 Installing SALOME _CFD or *Code_Saturne* and SALOME separately

The quickest way to get started is to retrieve a SALOME _CFD installer from *Code_Saturne* website: www.code-saturne.org. Then install SALOME _CFD by executing the installer in a terminal.

Alternatively, each application, *Code_Saturne* and SALOME, can be installed separately with an option allowing to integrate *Code_Saturne* inside Salome. Access to the applications is provided through their respective, dedicated internet sites. Install utilities, installation instructions, environment variables setup for Linux installations, and code documentation including user manuals can also be downloaded from the sites.

To integrate the *Code_Saturne* and the CFDStudy module in SALOME, SALOME must be installed first, followed by *Code_Saturne*:

1. SALOME

SALOME can be downloaded from the website: www.salome-platform.org/

Complete versions come with the source and binary files and interactive installation procedures.

2. Code_Saturne and CFDStudy module

Code_Saturne can be downloaded from the website: www.code-saturne.org

The code is packaged with all its libraries and the installer, which can also be downloaded from the same site, makes the installation process fully automatic. However, external, required packages, such as Python, are not part of the *Code_Saturne* package and must be acquired from their respective publishers and installed separately. As part of the installation process, the *Code_Saturne* installer will check for their availability and issue notifications if they are not found.

To enable the CFDStudy module, the *--with-salome* flag must be added as a parameter to the *configure* command. The flag must point to the path of your SALOME application.

For example: --with-salome=/opt/salome_8.1.0.

Additional information can be found in the installation guide of *Code_Saturne* available on the *Code_Saturne* website.

1.4 Getting Started with End-to-End SALOME/ Code_Saturne CFD Simulations

The process of creating the required *Code_Saturne* directory and file structures is decribed first, followed by an example for the Shear Driven Cavity Flow case.

File Structure

Code_Saturne makes use of a pre-set directory structure to access and save input and output files. Simulations are organised in terms of studies and cases. Conceptually, a study contains a series of cases which all rely on the same geometry. The cases represent different instances of simulations for this common geometry, for example for different operating conditions. The pre-set *Code_Saturne* directory structure can be easily created using the CFDStudy module within SALOME, as described below.

To set up a study and a case, first open the combined interface of *Code_Saturne* and SALOME by typing the command:

\$ code_saturne salome

in your terminal. This will start the SALOME platform with the CFDStudy module enabled.

Then, activate CFDStudy, either by selecting it in the module chooser in SALOME's top tool bar or by clicking on its icon (Figure II.1).



Figure II.1: Opening CFDStudy module

The module's activation screen will open as shown in Figure II.2.

	Activate module	
0	You're activating module CFDSTUI Please, select required action by pressing the corresp)Y . oonding button below.
New	<u>O</u> pen <u>C</u> onnect Load <u>S</u> cript	<u>C</u> ancel

Figure II.2: CFDStudy activation screen.

In the module's activation screen, select New to create a new study. Then enable **CFDSTUDY** toolbar, if it does not appear, in View Toolbars CFDSTUDY. In the toolbar or in the **CFDSTUDY** menu, click on Choose an existing CFD study or Create. The study and case creation panel will then pop up (Figure II.3).

<u>L</u> oad existir	ng study
☑ Create <u>S</u> tu	dy
Study location	/home/A41771
Study <u>n</u> ame	EXAMPLE_STUDY
Cases	
CASE1	
Ode_Satu	rne
copy from e	existing case
Coupling with the second se	ith Syrthes
	<u>●</u> <u>C</u> ancel <u></u> <u></u> <u>O</u> K

Figure II.3: *Code_Saturne* Study and Case Specification. Here a study called EXAMPLE_STUDY, containing CASE1, is created

In the panel, choose where you want the study to be stored on your computer and enter the names of the study and the first case. Further cases can be added to any study at a later stage. Leave the **create MESH directory** and **create POST directory** tick-boxes selected, as this will automatically create the entire directory structure required by *Code_Saturne*. Press **OK** to create the study and the case. The *Code_Saturne* directory tree for the study and case will then become visible in the Object Browser, as shown in Figure II.4.



Figure II.4: Case directory and file structure.

The meaning of the different directories and files is recalled here briefly. For detailed information, please refer to the *Code_Saturne* user manual [3].

- 🗂 DATA directory
 - $\circ \boxdot \texttt{REFERENCE} \ directory$
 - \rhd Tabulated thermophysical properties for chemical species and atmospheric properties
 - \triangleright cs_user_scripts.py file
 - 4 Python file in which users can modify Code_Saturne parameters and settings
 - \circ SaturneGUI file

 \triangleright Shell file pointing to the *Code_Saturne* GUI executable

• \boxdot RESU directory

 \circ Initially empty, it will contain the results files and $Code_Saturne$ outputs of each $Code_Saturne$ run, organised in chronological order

• 🕾 SCRIPTS directory

 \circ Bash file <code>runcase</code>, in which the <code>PATH</code> to the <code>Code_Saturne</code> executable and the run command are automatically setup for the case name

- \boxdot SRC directory
 - \circ User coding source files
 - $\circ \boxdot \texttt{REFERENCE} \ directory$
 - \triangleright Templates for all the available user coding files
 - $\circ \boxdot$ EXAMPLES directory
 - \rhd Practical examples of implementation of user coding for different purposes

Initially, the \bigcirc MESH and \bigcirc POST directories are both empty. By default when run from a case directory, *Code_Saturne* is set up to look for the mesh file in the \bigcirc MESH directory. The \bigcirc POST directory could contain post-processing routines if required.

Example: Creating the Shear Driven Cavity Flow Study

Follow the steps described in 1.4 to open CFDStudy and create a new study.

Then select the path where the study directory will be created and name the study **Driven_Cavity**. Tick the box Add new case(s) and enter the case name **Reynolds1000** as shown in Figure II.5. The boxes to create the \bigcirc MESH and \bigcirc POST directories have to be ticked.

<u>L</u> oad existin	g study
✓ Create Stud	iy
Study location	/home/A41771
Study <u>n</u> ame	Driven_Cavity
Cases	
Reynolds10	00
 Code_Satur 	ne
copy from e	xisting case
Coupling with the coupling with the coupling with the coupling of the coupling with the coupling wi	th Syrthes
	<u></u> ⊆ancel <u>≪</u> <u>O</u> K

Figure II.5: Specifying the Shear Driven Cavity Flow Study and Reynolds1000 Case

The whole directory structure of *Code_Saturne* is generated and can be visualized in the Object Browser tab (Figure II.6).



Figure II.6: Shear Driven Cavity Flow Study and Reynolds1000 Case file structure

The case is then ready to be set up, as described in 3.1.

Part III

Shear Driven Cavity Flow Case with CFDStudy

1 General description

In the second part of the tutorial, the preparation, simulation and analysis of the **Reynolds1000** case of the **DrivenCavity** study is described, from the construction of the computational domain and mesh to the preparation, running and post-processing of the CFD simulation. All the following steps in this second part are presented using CFDStudy module. However you could also go through this tutorial using *Code_Saturne* and SALOME separately.

1.1 What You Will Learn

Through this tutorial, you will learn how to perform an end-to-end CFD simulation using *Code_Saturne* [2, 3] together with the SALOME [1] platform, from creating the computational domain to analysing the end results and comparing them with available data. Specifically, you will:

- Create a computational domain with SALOME using available shapes and groups
- \bullet Create an hexahedral mesh with different mesh refinements in the $X,\,Y$ and Z directions with SALOME
- Setup a steady-state, viscous, laminar, isothermal, constant properties fluid CFD simulation with non-slip walls, a moving wall, and symmetry planes with the CFDStudy module and using the SALOME mesh
- Control and run the *Code_Saturne* simulation from the CFDStudy module
- Examine the *Code_Saturne* output and results files, including data along specified profiles and at monitoring points
- Analyse and visualise the results with SALOME

1.2 Case Description

A viscous fluid is contained in a box, or cavity. All the walls of the cavity are stationary, but for one wall which is sliding in its plane and sets the fluid in motion inside the box through entrainment. This is a well-know academic case for which published results are available [4].

For the purpose of this example, the cavity is rectangular, two-dimensional, and the top wall of the cavity slides from left to right, as illustrated in Figure III.1 below.



Figure III.1: Schematic of the 2D Lid Driven Cavity.

Geometry

The cavity is a square of length L = 1.0 m.

Fluid Properties

The fluid is given the properties listed in Table III.1 below:

Property	Value	Units
Density (ρ)	1.0	kg/m^3
Viscosity (μ)	10^{-3}	Pa.s

Table III.1: Fluid Properties.

Boundary Conditions

The problem is considered to be isothermal and the domain is fully enclosed by impermeable, non-slip walls. This means that, exactly at the surface of the walls, the fluid inside the box attaches to the walls and has exactly the same velocity as the walls. Therefore, the problem is fully defined by specifying the velocity of the walls (Table III.2). The side and bottom walls are stationary. The top wall slides in the positive X direction at a speed of $1.0 \text{ } m.s^{-1}$.

	Velocity component	
Surface	$V_x \ (m/s)$	$V_y \ (m/s)$
W_l	0.0	0.0
W_b	0.0	0.0
W_r	0.0	0.0
W_t	1.0	0.0

Table III.2:	Wall	velocities.
--------------	------	-------------

Flow Regime

The flow Reynolds number is evaluated to determine whether the flow is in the laminar or turbulent regime. The Reynolds number is calculated based on the lid velocity, the box dimensions, and the fluid properties.

For this case, the fluid properties and lid velocity have been chosen so that $Re = \frac{\rho \cdot U \cdot L}{\mu} = 1000$, indicating that the problem is in the laminar flow regime.

1.3 Creating the *Code_Saturne* **case**

The **DrivenCavity** study and **Reynolds1000** case are created by following the instructions in section II of this tutorial.

The next paragraphs describe how to setup and run the lid driven cavity simulations for a flow Reynolds number of 1000.

2 Creating the Computational Domain within SALOME

First you need to create the geometry of the cavity which defines the bounds of the computational domain and then mesh this domain. Installation of this tool is described in section II.

Code_Saturne is a 3D code and, therefore, the computational domain will be created as a 3D box, aligned with the X, Y, and Z directions. However, as we are only interested in 2D modelling and the flow motion in the X - Y plane, the box will be designed to be thin and contain only one computational cell in the Z direction.

2.1 Geometry

Once the CFD case structure is created from the CFDStudy module, the current SALOME file is saved by accessing File Save. For this tutorial, we choose the file name Cavity (Figure III.2). The file Cavity.hdf will be created in directory Driven_Cavity.



Figure III.2: Saving the SALOME file

It is advisable to save the file at regular intervals when preparing a case, preferably, before and following all significant modifications. That way your latest work is preserved and, should mistakes happen, it is always possible to exit SALOME and restart it from the latest saved file.

As SALOME is an integrated interface to different capabilities: CAD, meshing, visualisation, the next step is to access the **Geometry** module of the program in order to start building the shape of the computational domain.

In the main menu bar, either click on the 🔯 icon or choose **Geometry** from the drop-down module menu as shown in Figure III.3 below.



Figure III.3: Selecting **Geometry** in the module chooser.

The **Geometry** menus then become available in the top bar and the **OCC scene viewer** opens in the right hand side, where the geometry objects can be visualised and manipulated. The scene viewer comes with its own set of 3D-viewing menus to control the display and perform operations such as panning in and out, and rotating and moving the object in the scene.

The **Geometry** module of SALOME makes it possible to build bespoke objects from scratch from constituting elements such as vertices, edges and faces, or from pre-set generic shapes.

In this instance, we use the latter, more direct method and create the cubic cavity shape by selecting: New Entity Primitives Box. Alternatively, use the icon short-cut on the menu bar.

A Box Construction pop-up menu is then activated, as shown in Figure III.4 below.

File Edit View New Entity Operation	: Repair Inspection Tools Window Help netry V 🔽 🕱 🛪 📶 🥱 🚛 🥦 🚘 🚱 🛹 🕅 🐲	SALONE 8
• ✓ ⊙ ds つ ∿ ✓ ♡ ✓ • ↓ ⊕ ∄ 4 4 20 00 ✓ Object Browser Ø 8	∧ ∅ </td <td>🐿 🧻 🗋 »</td>	🐿 🧻 🗋 »
Name GEDSTUDY Geometry		🖻 🍕 🐣 🔹
 	Box Box Result name Name Cavity Dimensions At Origin Dx: 1 Dy: 1 Dz: 0.01 Close Help	
Python Console		0 8

Figure III.4: Box construction menu.

In this menu, choose the second method of Box construction, which makes it possible to build a prismatic box attached to the origin by defining its three dimensions. The default name is changed

to **Cavity** and the dimensions in the X, Y, and Z directions are specified according to the problem definition in paragraph 1.2.

The resulting Cavity object is shown in Figure III.5 below, as wireframe (top) and after applying shading (bottom).



Figure III.5: Wireframe (left) and Shading (right) display modes.

As shown in Figure III.5 (bottom), to change the Display Mode to Shading, right click in the scene viewer to open the drop down menu, then select Display Mode and Shading. Likewise, Change background ... may also be accessed from the same drop-down menu to change the background colour of the OCC viewer to light blue as in the figures.

At this point, the cavity object is complete and could be meshed straight away. However, since the box has been created as a whole entity it would then neither be possible in the mesher to specify different meshing characteristics for the X, Y and Z directions nor to identify the 6 external surfaces

as boundaries for the *Code_Saturne* calculations. Before meshing, it is then necessary to create and identify these elements. In the next step, you will automatically create **Edges Groups** using the **Propagate** utility. **Faces Groups** will then be created by manually selecting them.

With the **Cavity** object selected in the Object Browser in the left-hand column, in the top menu bar, click on Operations Blocks Propagate as shown in Figure III.6 below. By using the Apply and Close action at the bottom of the menu, you will be able to automatically create three Edges Groups, each of them containing every edges parallel to one direction.



Figure III.6: Using the **Propagate** utility to automatically create **Edges Groups**.

One can now rename them as X_Edges, Y_Edges and Z_Edges by right-clicking on each group and then select Rename. The Cavity object in the object browser should now contain three Edges Groups as shown in Figure III.7 below.





Figure III.7: Cavity with the **X_Edges**, **Y_Edges** and **Z_Edges** groups.

The next step is to group faces according to the boundary types listed in Table III.2 above. Right-click on **Cavity** in the object browser and select **Create Group** to activate the pop-up menu, as shown in Figure III.8 below.



Figure III.8: Creating the **Stationary_walls** face group.

Alternatively, access the group creation via New Entity Group Create Group.

Select the **m** icon as **Shape type**, name the group, and ensure that the **Main Object** is **Cavity**. To be easily identifiable, the groups are named after their boundary types. The first group is created for the external, stationary walls and named **Stationary_walls**. With the default **No restriction** choice already active in the **Main Shape Selection restriction** menu, start adding the required faces by

left-clicking on them in the OCC viewer and then pressing Add. One can zoom with the mouse wheel as well as rotate the Cavity object by holding Ctrl + Right Click to select the faces more easily. The faces may be added one by one, or together by multiple-selection through left-clicking and holding the Ctrl key. Once the three faces are selected, one can check them by pressing the Show only selected button. The Create Group menu should look as shown in Figure III.9.



Figure III.9: Selected faces of the **Stationary_walls** face group.

Press Apply and repeat these steps for the next two groups: Symmetry_plane and Sliding_wall, pressing Apply each time. Conclude the creation of face groups by pressing Apply and Close.

The face and edges groups are now listed under the Cavity object in the Object Browser (Figure III.10).



Figure III.10: Cavity with the edges, faces, and groups.

Cavity is now ready to be meshed. Save the file and proceed to Meshing.

2.2 Meshing

Switch to the meshing module, which opens the **VTK scene viewer** as shown in Figure III.11 below. In this view, the background colour has been changed to white similarly to the **OCC scene viewer** for the **Geometry**, by right clicking in the viewer and accessing **Change background**



Figure III.11: Switching to the **Mesh** module.

Similarly to the earlier creation of the geometry object **Cavity** in the **Geometry** module, a **Cavity_mesh** mesh object is created first by selecting Mesh Create mesh in the main menu bar.

As shown in Figure III.12 below, in the **Create mesh** pop-up menu, type **Cavity_mesh** for the **Name** and select **Cavity** as the **Geometry** object by left clicking on it in the Object Browser.

rt Browser BLB	8 : 99 N : 199 S3 M : 10 12 : 10 9 13 14 14 12 : 17 17 17 17 14 15 14 15 14 15 14 14 14 14 14 14 14 14 14 14 14	-
Nome © COSTUP © Sonstry + 0 / OX / OY / OZ / OX / OY / OZ / OZ <th>ULC LEARNE - VANNE - A DUC LEARNE - VANNE - A Create mesh Name Create mesh Create mesh Cavity Mesh type Any Hoometry Covity Hexat type Any Hoometry Covity Hexat type Any Hexat type Any Hexat type Any Hexat type Any Covity Mesh Star Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis</th> <th>30</th>	ULC LEARNE - VANNE - A DUC LEARNE - VANNE - A Create mesh Name Create mesh Create mesh Cavity Mesh type Any Hoometry Covity Hexat type Any Hoometry Covity Hexat type Any Hexat type Any Hexat type Any Hexat type Any Covity Mesh Star Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis Add Hypothesis	30

Figure III.12: Creating the main mesh object.

The main mesh object holds the global characteristics of the Cavity mesh.

For the volume, in the **3D** tab choose **Hexahedron** (i,j,k). All the (external) surfaces of the Cavity will be meshed similarly, therefore a global setup is applied for the entire mesh via this pop-up menu by clicking on the **2D** tab and selecting **Quadrangle:** Mapping. These two selections will result in a block-structured mesh, with an (i, j, k) matrix structure. This requires identical discretisations along all the edges aligned along the same directions. If we wanted the discretisation to be the same along all the directions, for example 15 sub-divisions along the X, Y, and Z edges, the **1D** tab could be selected and this unique discretisation specified. However, here we need to be able, at least, to specify the Z direction independently of the two others to create a 3D domain which is only one cell deep. Therefore, the edge discretisations will be specified separately using **sub-meshes**.

The sub-mesh defining the meshing characteristics in the X direction is specified first.

Right-click on **Cavity_mesh** in the Object Selector and select **Create sub-mesh**. Alternatively, with **Cavity_mesh** selected (highlighted) in the Object Browser, the sub-meshes can be created from Mesh or the short-cut icon in the main menu bar. This activates the **Create sub-mesh** pop-up menu in the **VTK scene viewer** as shown in Figure III.13 below.

	Create sub-mesh				
Name	X_mesh				
Mesh 🖉	Cavity_mesh				
Geometry 🕜	X_Edges		. н.	mothesis Construction	
Mesh type	Any	~) _ ~)	With the second seco		
Algorithm	Wire Discretisation	×	Name	X_discretisation	
Hypothesis	<none></none>	V 20 20	Number of Segments	50	^ V
Add. Hypothes	is <none></none>	V 20 20	Type of distribution	Equidistant distribution	~
		4	<u>O</u> K <u>C</u> ancel		Help
	Assign a set of hypotheses	~			
Apply and Close	e <u>A</u> pply C <u>l</u> ose	Help			

Figure III.13: Creating the X_mesh sub-mesh.

In the name field, enter X_mesh. The Cavity_mesh should already be selected in the Mesh field. If not, select it by clicking on Cavity_mesh in the Object Browser. Next, pick X_edges for the Geometry field which stores the Geometry Objects the sub-mesh will apply to. Selecting an edge group for the Geometry automatically disables the 3D and 2D tabs in the Algorithm menu. Select Wire discretisation in the drop-down Algorithm menu and press on the gear button next to Hypothesis to specify the discretisation in its own pop-up menu. The Number of Segment object is renamed X_discretisation for future reference. Choose 50 segments with an Equidistant distribution and press OK. Returning to the Create sub-mesh menu, press Apply to instantiate the X_mesh sub-mesh. The sub-mesh objects for the other two directions are defined similarly.

Starting with the Y direction, the sub-mesh is renamed Y_mesh . The same number of segments is applied in the Y and X directions. Therefore, for this sub-mesh the previously created $X_discretisation$ Number of Segment Object is reused in the Hypothesis. Complete the specification of Y_mesh and press Apply.

Lastly, the sub-mesh for the Z direction is defined. Named **Z_mesh**, it is similar to the **X_mesh** and **Y_mesh**, but defines the discretisation as 1 segment only. For **Z_mesh**, define a new **Number of Segment** object named **Z_discretisation** and specifying 1 segment. Complete the specification of **Z_mesh** and press [Apply and Close] to instantiate **Z_mesh** and leave the **Create sub-mesh** menu.

The mesh characteristics of the **Cavity** are now fully specified, as shown in Figure III.14 below where all the characteristics of **Cavity_mesh** and its sub-meshes have been listed.



Figure III.14: Complete specification of Cavity_mesh and sub-meshes in the three directions.

To mesh the computational domain, right-click on **Cavity_mesh** in the Object Browser and select **Compute**, as shown in Figure III.15 (left) below. Again, with **Cavity_mesh** selected, **Compute** can also be accessed via the main menu bar <u>Mesh</u> Compute or via the short-cut icon **Solution**. The mesh is created and a pop-up window appears automatically summarising the mesh characteristics (Figure III.15 (right)).



Figure III.15: Computing the mesh (left) and resulting mesh (right).

Before the mesh can be used in *Code_Saturne*, the boundary surfaces must be identified so that it will be possible to impose boundary conditions in *Code_Saturne*. To designate these surfaces, we now can call on the groups of faces which were created previously in the **Geometry**.

Right-click on **Cavity_mesh** in the Object Selector and select **Create Group from Geometry** to create containers of mesh elements. The **Create Group from Geometry** pop-menu is activated. Left click on **Stationary_walls**, **Sliding_wall** and **Symmetry_planes** in the **Geometry** part of the Object Chooser while holding **1**. The three groups will be automatically added in the **Elements** box as shown in Figure III.16 below. Ensure that **Cavity_mesh** has been selected for the mesh and press **Apply and Close**].



Figure III.16: Creating the mesh face groups.

To help visualisation, the colour of these groups of faces can be switched to different colours for each group. Right click on the name of the group in **Groups of Faces**, select **Edit group** and choose the colour.

The groups containing the boundary faces are now defined and may be visualised in the **VTK scene viewer**, together or individually, by selecting the groups of mesh faces in the Object Browser. For example, the **Stationary_walls** group is shown in Figure III.17 (left), and all the groups are visualised in Figure III.17 (right). Note that the names of the groups which have been selected appear in the top right corner of the viewer, in the colour scheme chosen for the group.



Figure III.17: Visualing the **Stationary_walls** group on its own (left) and all the groups (right).

The mesh is now ready to be exported in a separate file which can be read in by *Code_Saturne*.

Before exporting the file, it is important to highlight an important naming convention, as the names of the mesh and its elements will be used as their identifiers in *Code_Saturne*.

Save the SALOME file and export the mesh file in .med format by selecting from the main menu: $\boxed{\text{File}}$ $\boxed{\text{Export}}$ $\boxed{\text{MED file}}$, as shown in Figure III.18 below. The file should be placed in the $\boxdot \text{MESH}$ directory of the DrivenCavity study, where *Code_Saturne* will expect the file to be situated by default.



Figure III.18: Exporting the mesh file in .med format.

For the file name, choose **Cavity_mesh**; the .med extension is automatically added. You are now ready to set up the CFD simulation with the CFDStudy module.

3 Simulating within SALOME

3.1 Setting up the CFD Simulation

To set-up the CFD Simulation, select the CFDStudy module by clicking on the CFDStudy button or by selecting CFDStudy in the modules drop-down menu as shown in Figure III.19.

Y	SALOME		
63	Geometry		
S.	Mesh		
ų,	CFDSTUDY		
S	SYRTHES		
///	ParaViS		
7	MED	SCFDSTUDY V	& 🗟 🛪 📶
	YACS	0 x	
9	JobManager		VIK
-	Homard		
R	HexaBlock		
San.	GenericSolver		

Figure III.19: Selecting the CFDStudy module.

the CFDStudy modules by clicking on Launch GUI as shown in Figure III.20 below.

۲	Name	: 65		
	□ S CFDSTUDY □ ♥ ♥ Oriven_Cavity I □ □ MESH □ □ POST	1		
	■ Reynolds1000 ⊕ DATA Control Contro	Cur Launch GUI X Remove Refresh	F5	
	 	Expand All Collapse All Find	Ctrl+F	

Figure III.20: Launching the Code_Saturne GUI inside the CFDStudy module.

Once the *Code_Saturne* GUI is open, it is advisable to start by saving your CFD set-up file. In the CFDStudy tool bar, click on the **Gate CFD Data File** button (Figure III.21).

B	Sitte	22	а П	Gui	M		-	0
					 Save C	FD Da	ata File	

Figure III.21: Button bar of CFDStudy module.

By default, the file will be saved in the CDATA directory. Name the file Reynolds1000.xml.

deaths and noths	VTK scene	4 - viewer1	8		
Colculation environment	50 C 17 18 18	B B S P P P P P + + + + + + + + + + + + + +		Directory of the case	
Colculation central.				our rest random server of	mercennynnynen 📼
				Associated sale-directories to	d size same
		Save File AL		Data	DATA
	Non:	Reynolds1000.xm		Results	RESU
	Ferrard and the description	. C Tatala Dava Cara Remitation Di	a community	User sabroutines	SRC
	Lingsine des di dese		- Crementonion	Distant and the	COMP.
	Barrancia	Nam	- Talla Hadfil	Post ming scripts	2.4713
	Q Rectexter	BEFERENCE	Hera1054		
	O Ricemmert u.				
	E DATA				
	SH TH17180N				
	St Bureau				
	El Système de fu-				
	Si Commente				
	Musican				
	an instruction				
	the images				
			Code.Saturne GUI fées		

Figure III.22: Saving the Code_Saturne case.

You can now proceed with setting up the case, in the top down order of the folders in the left hand column of the CFDStudy module.

Press on Calculation Environment to open the folder and select Meshes Selection in order to specify the mesh file which will be used for these calculations. In the Meshes tab, for Mesh import, Import meshes is already selected by default. The Local mesh directory should already be pointing to the \square MESH directory for the study, where the mesh file had been exported in paragraph 2.

Initially, the **List of meshes** table is empty. Click on the + sign button below the table and a popup menu will show the content of the local mesh directory (here \bigcirc . . /MESH) from which the mesh file can be selected. Select the Cavity_mesh.med file previously exported from SALOME at the end of the meshing process, and press Open. The file name will then be added to the list of meshes (Figure III.23).



Figure III.23: Selecting the mesh file for the calculations.

No further input is necessary as the faces do not require reorientation, joining, or sub-dividing, and there are no periodic boundaries to be concerned with.

Next, move to **Thermophysical Models** to specify the flow physics for the calculations.

In **Calculation features**, change **unsteady flow** to **steady flow** in the choice of algorithms at the top. Leave all the other default values unchanged: single-phase flow is active and the combustion and electrical models are all inactive (Figure III.24).

steady flow	~
ulerian-Lagrangian multi-phase treatment	
off	~
tmospheric flows	
off	~
Sas combustion	
-11	
on	Ÿ
ulverized fuel combustion	V
Pulverized fuel combustion	*
Pulverized fuel combustion	· · · · · · · · · · · · · · · · · · ·
Pulverized fuel combustion off ilectrical models off	v
Pulverized fuel combustion off Electrical models off Compressible model	v v
Pulverized fuel combustion off Electrical models off Compressible model	v v
Durbried fuel combusition off Electrical models off Compressible model	v v v
Durverized fuel combustion off Electrical models off Compressible model off	v v

Figure III.24: Selecting the flow physics.

Continue setting up the **Thermophysical Models**.

By default **Mobile mesh (ALE method)** is not selected. As the mesh in our case is stationary, **Mobile mesh** does not need visiting and changing. However, **Turbulence models** needs to be changed to **No model (i.e. laminar flow)**, which is the correct flow regime for the flow Reynolds number that we have chosen in paragraph 1.2 (Figure III.25).

rbulence model	
No model (i.e. laminar flow)	~

Figure III.25: Selecting the laminar model.

Thermal model is inactive by default. As our case is isothermal, it does not need to be visited and changed.

Save the file and, as there are no additional scalars, skip the **Species transport** category and move to the **Physical Properties** folder where the thermal and physical characteristics of the fluid of interest for the calculations will be specified.

In **Reference values**, the pressure reference value of 101325.0 *Pa* is set by default and does not need to be changed. The other default values for velocity and length can also be left unchanged. However, the **Fluid Properties** must be modified for our fluid. Click on **Fluid Properties** in the left hand panel and, as per Figure III.26 below, alter the values of **Density** and **Viscosity** to the chosen values which are listed in Table III.1 above.

	material	user material 🗸	
	method	user properties 🛩	
	reference	user_material	
Density	constant 🗸	8	
	Reference value	ρ 1	kg/m³
Viscosity	constant v	12	
	Reference value	μ 1.0e-3	Pa.s
Specific heat	[B 2	
	Reference value	Cp 1017.24	J/kg/K

Figure III.26: Specifying the fluid properties.

The default value for **Specific heat** does not need to be altered as it is not a parameter in these isothermal calculations. Likewise, gravity is not a parameter of this problem (constant density), therefore the **Gravity** category does not need to be visited and changed.

Click on the **Volume Conditions** folder. Sub-volumes of the computational domain (**Volume regions definition**) have not been defined as separate regions for initialisation and **Coriolis source terms** are not taken into account in this tutorial. Therefore, the **Volume Conditions** category must be changed only for the **Initialization**.

In this sub-folder, press on the icon we next to **Velocity** to bring up the **Mathematical Expression Editor** and change the initial first component of the velocity to 0.1 m/s as shown in Figure III.27 below.



Figure III.27: Specifying the initial values.

To define both the boundaries and the boundary conditions, under the **Boundary Conditions** folder, select **Definition of Boundary Regions**.

On the left side of the screen move from Driven_Cavity.Reynolds1000.Reynolds1000.xml Browser

		Code_Saturne
EDF R&D	Code_Saturne version 5.0 tutorial: Shear Driven Cavity Flow	documentation Page 28/47

to **Object Browser**. Then in the **Mesh** part click on the group of faces desired and on the right side of your screen press on Add from Salome. The group added by default is set with the boundary condition **Wall** as shown in Figure III.28.



Figure III.28: Importing groups of faces from SALOME.

In case there are a lot of groups, it is quicker to do a **Mesh quality criteria** calculation. Select the **Preprocessing Mode** as shown in Figure III.29 and go to **Calculation management**, then select **Mesh quality criteria** in the **Run Type** category.

Calculation Mode							
		M	×	(¢		
Preprocessing Mode							

Figure III.29: Preprocessing and Calculation modes.

This can be used to analyse the mesh. For visualisation of the mesh quality criteria, export the files in the specified output format. To do so, go to **Calculation control** and select **Output control**. Then, in the **Writer** tab, select the desired output format for the mesh quality criteria. As shown in Figure III.30 below, the **EnSight** format is preselected by default. Doing a **Mesh quality criteria** also enables *Code_Saturne* to detect the different groups of faces in the mesh.



Figure III.30: Selecting the mesh quality criteria output file format.

Go back to Calculation Mode and to the Boundary Conditions tab, select Definition of Boundary Regions and then press Import groups and references from Preprocessor listing. The Select a Code_Saturne Preprocessor listing pop-up frame will then appear, which allows you to select the preprocessor.log file to use to load the boundary surfaces of the computational domain. You will then be able to apply specific boundary types to each of the surfaces or groups of surfaces. This file is in a directory called cpreprocess20161020-0817 for instance. In the pop-up frame, select the file preprocessor.log as shown in Figure III.31.

Elle Edit View CFDSTUDY Tools Window He	elp				SF	ILOME 8
Driven_Cavity.Reynolds1000.Rey 🛞 🗷 CFDSTUDY	scene:4 - viewer:1	0CC scene:2 - viewer:1	_≌ *s &s ⊌ ™ ⊠]	Driven_Cavity.Reynolds100	0.Reynolds1000.xml	0 8
Mentizy and paths GetCulation environment Bentizy and paths GetCulation environment Demroophysical models Demroophysical models Demroophysical models Demroophysical models Demroophysical models Demroophysical models GetCulation environment GetCulation environment CetCulation environment CetCulation environment	Kanada - Vanari Recourds Recourds Recourds Recourds Recourds Recent. S Récemm. S AtA S H2788. S Systems. S Systems.	OCC seeme 2 viewer:1 Select a code. Sature Tutorials OrienCavity Nom performance log performance log setup.log	Stern Preprocessor Bitty Reynold:1000 RESU preprocess2	Perfectation versions of the soundary re- Label Zone Nature Inc Animal Soundary re- Inc Animal Soundary re-	d Delete Add from Salame Istino ry, wals d Delete Add from Salame Isting and references from Preprocessor lating	
U Driven_Cavity.Reynolds1000.R						

Figure III.31: Importing boundaries from the preprocessor.log file.

All the boundary faces defined in the mesh are now loaded and are set by default with the **Wall** boundary condition.

The first boundary group in our problem contains the side walls, which are considered stationary and non-slip. In the first row of the list of boundary regions, double click the cell in the Selection area column in order to identify the boundary region in the mesh file. With the cursor in the cell, type in the exact name of the boundary in the mesh file, i.e., Stationary_walls. If the preprocessor.log

file has been used, the **Selection area** is already completed.

The second set of boundaries are the symmetry planes which are used to enforce the 2D nature of the problem. In the second row, double click on Wall in the Nature column to activate the drop down menu of boundary types. Choose Symmetry and release the menu. In the Selection criteria cell, input the name of the boundary in the mesh file, i.e., Symmetry_planes, if no preprocessor.log file has been used.

In the third row of the table, as for the two other boundary conditions, input the name of the sliding wall boundary in the mesh file, i.e., **Sliding_wall** if no preprocessor.log file has been used.

Note that it is better to press enter after typing in the name of the region to ensure that it is properly recorded by the GUI.

The boundaries and their global types are now defined, as shown in Figure III.32 below.

De	Definition of boundary regions							
	Label	Zone	Nature	Selection criteria				
	BC_1	1	Wall	Stationary_walls				
	BC_2	2	Wall	Sliding_wall				
	BC_3	3	Symmetry	Symmetry_planes				
	Add Delete Add from Salome							
	Add fror	n Prep	processor list	ting				
	Import groups and references from Preprocessor listing 🛛 🔚							

Figure III.32: Boundary definition.

Having defined their type, the exact characteristics of the boundaries must be further specified. Click on the **Boundary conditions** sub-folder which presents the list of boundaries which can be further specified (Figure III.33).

Bo	undary conditio	ons		
	Label	Zone	Nature	Selection criteria
	BC_1	1	wall	Stationary_walls
	BC_2	2	wall	Sliding_wall
	Smooth or rou	ugh wall		
		⊙ sn	nooth wall 🔘 r	ough wall
		л		
	Stiding wa			
	U 1.0	m/s	V 0.0	n/s W 0.0 m/s

Figure III.33: Specifying the sliding wall.

The symmetry boundaries are fully defined and do not need further specification. Therefore, they do not appear in the list. However, **Wall** boundaries can each be further defined as **smooth** or **rough**, and as **Sliding wall**. By default, the wall surface is **smooth** and this parameter does not need to be changed. The **Stationary_walls** boundary is fixed and fully specified. However, click on the **Sliding_wall** boundary and activate the **Sliding wall** selection. Fields then appear for the **U**, **V**, and **W** velocity components of the wall. By default, these velocities are null. Click in the **U** field and enter 1.0, as shown in Figure III.33.

The mesh and physics of the problem have now been set up. Now, parameters related to the calculation can be specified.

In Numerical Parameters, leave the settings unchanged in the Global parameters sub-folder and move to the Equation parameters. In the Scheme tab, apply the Centered scheme and disable the Slope test for the velocity (Figure III.34). The former will provide second-order accuracy in space and the latter is unnecessary for a smooth flow.

Solver Schen	ne Clipping				
Name	Scheme	Blending Factor	Slope Test	Flux econstructio	RHS Sweep econstructic
pressure					2
velocity	Centered	1		Z	1

Figure III.34: Specifying the flux schemes.

To set the duration of the run and the time step, move down the selection tree to the **Pseudo-Time step** sub-folder. Change the **Number of iterations** to 400 and increase the **Maximal CFL number** to 8.0, leaving the other parameters unchanged (Figure III.35).

Time step option		
Reference time step	0.1	s
Number of iterations (restart include	d) 400]
Maximal CFL number	8.0	
Maximal Fourier number	10.0	
Minimal time step factor	0.1	
Maximal time step factor	1000.0	
Time step maximal variation	0.1	
Option zero time step		

Figure III.35: Selecting the number of iteration and CFL number.

Move to the **Calculation control** folder. **Time averages** are not of interest for these simulations and the sub-folder does not need to be visited. However, we want to keep track of the solution at different monitoring points to see how it evolves during the calculations. Aside from the solution residuals and the minimum and maximum values of the flow variables which *Code_Saturne* outputs during the calculations, tracking the solution at significant monitoring points is a very important mean of gaining

confidence in the convergence of the calculations and judging whether calculations have been run for a sufficiently large number of iterations. This is explained further in paragraph 3.2 below.

Click on the **Output control** sub-folder panel. The first three tabs, **Output control**, **Writer**, and **Mesh** are set by default to the correct values for this case and do not need to be changed. The **Log frequency** in the **Output control** tab is set to print the calculations diagnostics such as the residuals at each time step. In the **Writer** tab, the format of the **results** file which will be used for post-processing is already set to **EnSight**, which is compatible with the SALOME module **ParaVis** which will be used to post-process the results after the run. The file will be located in the **Gpostprocessing** sub-directory of **GReynolds1000/RESU/runDateAndTime**, where **runDateAndTime** corresponds to the time at which the run was started. Clicking on the **results** row brings up the additional information about the **Frequency**, **Time-dependency** and **Options**. As already set by default, the **results** file will only be written at the end of the run and on the **Fixed mesh** used for this case. The **Options** relate to the specific details of the file format. The **Mesh** tab is already set to output the calculations data in all the fluids cells and at all boundary faces to the **results** file.

In the **Monitoring Points** tab, change the **Monitoring points output** file format to .csv so that the files may be read in ParaView. Using the + button, add four monitoring points located at (0.25, 0.5, 0.0), (0.5, 0.25, 0.0), (0.5, 0.75, 0.0) and (0.75, 0.5, 0.0) (Figure III.36).



Figure III.36: Defining the monitoring points.

A visualization of the monitoring points in the VTK Viewer in the middle of the screen, as shown in Figure III.36 above, is available if **Display monitoring points on SALOME VTK viewer** box is ticked.

Solution c	ontrol		
	Output label	Internal name	Post- processing
Stress		stress	
Stress	, normal	stress_normal	
Stress	, tangential	stress_tangential	
Yplus		yplus	

Figure III.37: Selection of output variables on the surfaces of the domain.

Click on the **Surface solution control** sub-folder and disable **Post-processing** for **Yplus** and **Stress** as they are not relevant to these simulations (Figure III.37).

Lastly, we want to output one-dimensional profiles of variables along straight lines at the end of the calculations. Click on **Profiles** and add two profiles which go through the centre of the Cavity by clicking on the $\boxed{\text{Add}}$ button. The first one for the X velocity along the Y axis and the second one for the Y velocity along the X axis. In turn, specify all the fields listed below the table of profiles, starting with **Filename** and finishing with the variables which are to be stored on output.

For the X velocity profile, choose XVel_YaxisCenterLine for Filename, and .csv for Format so that the profiles may be read in the ParaVis post-processing module of SALOME. The Output frequency must be set to at the end of the calculation. To define the line, press on the Mathematical expression editor button adjacent to Line Definition. The line is defined by the following equation :

$$\begin{cases} x = 0.5 \\ y = s \\ z = 0.0 \end{cases}$$

By definition, s varies between 0 and 1.0. For the Number of points enter 50 to account for the 50 cells across the domain's height. Finally, click on Velocity[0] and use the \rightarrow button to add it in the list of variables, and press Add to store the profile in the list (Figure III.38).



Figure III.38: Specifying the 1D output profiles.

Repeat the procedure for the Y-velocity profiles, this time entering **YVel_XaxisCenterLine** for **Filename**, and the following equation for the line defition :

$$\begin{cases} x = s \\ y = 0.5 \\ z = 0.0 \end{cases}$$

Select the **Velocity**[1] variable instead of **Velocity**[0]. The **Reynolds1000** case CFD simulation is now ready to run.

In the folder Calculation management, go directly to the Prepare batch calculation sub-folder. By default, the calculation-restart is disabled in Start/Restart and so this sub-folder does not need to be visited. In the **Prepare batch calculation** sub-folder panel (Figure III.39), select the script file which is by default **runcase** by clicking on the folder button **a**. A pop-up window opens up, select the **runcase** file and click on **Open**. For the **Run type**, select **Standard** (calculations without mesh or partitions import or pre-processing) and choose **1** processor for the **Number of processes**. Leave the **Run id** and **Threads per task** to their default value.





Press the Start calculation button to run *Code_Saturne*.

3.2 Running and Analysing the Simulation

Upon firing the *Code_Saturne* run from the GUI, confirmation that *Code_Saturne* is running, "*Code_Saturne* is running", appears in the window from which the GUI was started. This is followed by further messages indicating what stage the calculation is in, from "Preparing calculation data" to "Saving calculation results".

Wait for the calculations to complete and enter the \bigcirc Reynolds1000/RESU directory or open its contents via a browser to inspect its contents. Explanations of the meaning and purpose of the different output files and directories resulting from a run are available in the *Code_Saturne* Users Guide [3] and are not repeated here. Instead, we highlight individual items which relate to this specific run and how the output information should always be used in order to analyse a calculation.

The \bigcirc RESU directory now contains a new directory named after the date and time at which the calculation was started, expressed on a 24 hours clock in the format "YearMonthDay-HourMinutes".

In this latter directory, notice in particular:

- The profile files XVel_YaxisCenterLine and YVel_XaxisCenterLine, written in .csv format,
- The listing file,
- The monitoring and mostprocessing directories.

With your text editor, open one of the profile files, to inspect its structure. The requested variables listed in column format as a function of the (x, y, z) coordinates of the points along the profile line defined in the GUI.

Open the **listing** file to check that the calculation has converged towards stable values. The minimum and maximum values of the solution variables and the solution residuals for these variables are listed in summary tables at each iteration. By comparing the initial and final values in the **Drift** column for each variable, and their evolution throughout the calculation, check that the residuals have decreased by at least two orders of magnitude.

****** INFORMATION ON CONVERGENCE

	Variable	Rhs norm	N_iter	Norm. residual	Drift	Time residual
с с с с	Velocity Velocity[X] Velocity[Y] Velocity[Z] Pressure	0.24601E-03 0.18599E-03	24 22	0.46251E-02 0.58361E-08	0.10142E+00 0.10057E+00 0.85128E-03 0.82487E-29 0.10000E+01	0.25130E+02 0.90909E+01

Figure III.40: Code_Saturne Convergence history from the listing file, after 1 iteration.

****** INFORMATION ON CONVERGENCE

	Variable	Rhs norm	N_iter	Norm. residual	Drift	Time residual
c c c c c	Velocity Velocity[X] Velocity[Y] Velocity[Z] Pressure	0.28275E-03 0.39566E-10	18 0	0.10307E-07 0.67068E-07	0.67436E-15 0.31555E-15 0.35882E-15 0.55548E-44 0.67068E-07	0.13487E-06 0.12750E-15

Figure III.41: Code_Saturne Convergence history from the listing file, after 400 iterations.

Inspect one of the monitoring files in the \boxdot monitoring directory. The value of each variable is recorded in separate files at all the monitoring points defined in the GUI and as a function of iteration.

Finally, verify that the **RESULTS.case** file containing all the mesh and output variables information for post-processing has been output in the **mostprocessing** directory.

Having validated the calculation itself, you can now proceed to examining and post-processing the results by returning to the SALOME platform.

3.3 Post-processing the Results

In SALOME, select **ParaViS** from the drop-down module selector in the top menu bar. The name of the module will add itself to the Object Browser list and the ParaView-specific panels and menus will be activated, including a new **ParaView scene viewer** window.

Before loading the run data in ParaView, modify the default colour schemes. To visualise scenes on a white background, which is advantageous for printing, change the default settings for ParaView by clicking on File Preferences in the top menu. For ParaViS, in the ParaView Settings tab, change the colours to black for the foreground and text, white for the background, and grey for the surfaces,

as shown in Figure III.42 below. Press Apply to enforce the new settings and OK to validate your selection when you are satisfied with the changes.

SALOME	ParaView Settings ParaVis Settings						
Mesh	Search (use Esc to clear text)						
SYRTHES	SYRTHES General Camera Render View Color Arrays Color Palette						
M ParaViS	Color used when solid coloring surfaces and faces.						
YACS	Surface						
🔮 JobManage	Color used for rendering elements like wireframes, points.						
🔫 Homard	Foreground						
HexaBlock	Color used for the edges when using 'Surface With Edges' representation.						
GenericSol	Edges						
	Color used for text and other annotations.						
	Text						
	Selection						
	Color used for showing selected cells/points.						
	Selection						
	Color used for showing interactive selection.						
	Interactive Selection						
< — >	Background						
<u>o</u> k	Apply Defaults Import Close Help						

Figure III.42: Setting the colour preferences for ParaView.

The data to post-process can now be imported in ParaView.

First, you are going to load the monitoring point data to validate that a stable, steady-state solution has been obtained. From the top menu bar, select File Open ParaView File. In the Open File pop panel, navigate to the monitoring sub-directory and select three files. The multiple selection is performed by holding the Ctrl key down as you select the files. Select probes_Pressure.csv, probes_Velocity[X].csv, and probes_Velocity[Y].csv. Close the panel by clicking OK. The three sets of data are now displayed under their file name in the Pipeline Browser. For each file, press Apply to load the data. By default, the data is visualised in tabular form in the ParaView scene viewer. Close the View by clicking the cross button at the top, right hand corner of the view. Then, click on the D button to replace the data view by a Create View menu (Figure III.43).



Figure III.43: ParaViewCreate View menu.

Click on the Line Chart View button and prepare the line plots, starting with the pressure data. Make the data visible for that file by clicking on the eye symbol \bigcirc next to the file name in the **Pipeline Browser**. Go to the **Display** tab and set **Attribute Type** to **Row Data**. In the **Series Parameters** category, unselect the **iteration** variable. The variables 1, 2, 3, and 4, representing the pressure at each iteration at the four monitoring points you specified should be selected and the graphs of their evolution as a function of the number of iterations (time steps) should now be displayed in the view window. For clarity, change the legend for each variable. In the **Series Parameters** category, leftclick on the row for variable 1 to select it. Double left-click on the name of that variable, in the **Legend Name** column to edit it. Change the name of each variable to **Pressure (1)**, **Pressure (2)**, **Pressure (3)** and **Pressure (4)** (Figure III.44 and III.45).



Figure III.44: Specifying the line plot's legend.

Properties	0
<u>■</u> Apply	
Search (use Esc	to clear text)
📟 View (Line Cha	rt View)
Title	
Chart Title	Jse \${TIME} to display current time
Annotation	
🗹 Show Legend	
Left Axis	
Left Axis Title	ressure (Pa)
Left Axis Range	
🗆 Left Axis Log Sc	ale
🗆 Left Axis Use Cu	ustom Range
Bottom Axis	
Bottom Axis Title It	teration
Bottom Axis Range	
Bottom Axis Log	g Scale
Bottom Axis Use	e Custom Range

Figure III.45: Specifying the line plot's axes.

In the View window's tool bar Change the **Bottom Axis** title to **Iteration**, and the **Left Axis** title to **Pressure (Pa)** (Figure III.46). The graph shows that the pressure at the four monitoring points becomes constant after about 100 iterations (Figure III.46).



Figure III.46: Pressure at the four monitoring points as a function of iteration.

Repeat the same operation for the remaining velocity files. For both files, change the left axis title to **Velocity** ($\mathbf{m/s}$). For the X velocity, change the variable names to **X Vel** (i), where i = 1, 2, 3, and 4. Keep the default colours for each line so that they are consistent for each monitoring point. For the Y velocity, change the variable names to **Y Vel** (i), where i = 1, 2, 3, and 4. The graphs also confirm that the velocity components at the four monitoring points settle to a stable, constant value after about 100 iterations, indicating that the calculations are converged (Figure III.47).



Figure III.47: Velocity components (X (left), Y (right)) at the four monitoring points as a function of iteration.

You may now post-process the results from the calculations. From the top menu by selecting File Open ParaView File, or by right-clicking on the builtin object in the Pipeline Browser panel and selecting Open, access the pop-up Open File panel and choose the RESULTS.case file from your run. The RESULTS.case object will be added to the Pipeline Browser and its contents displayed in the Object Inspector panel underneath. Press Apply to load the data in ParaViS.

Next, to extract the computed data in terms of fluid domain and boundary data, in the top menu bar click on Filters Alphabetical Extract Block. The new object ExtractBlock1 now appears in the Pipeline Browser. With the object highlighted, move to the Block Indices panel of the Properties tab. In the data tree, under Root, select Fluid domain to visualise the data calculated at the cell centres inside the cavity walls, as shown in Figure III.48.

Properties (ExtractBl	ŝ
Block Indices	
E Root	
🗹 Fluid domain	
Boundary	
	2

Figure III.48: Extract the Fluid domain block in ParaViS.

Left click on the **Display** tab of the **Object Inspector**. In the category **Coloring**, two drop-down menus allow you to choose the variable to visualise. Choose **Velocity** in the first one and **Magnitude** in the second one, as shown in Figure III.49.

Coloring			
礡 Velocity	~	Magnitude	~
🔒 Edit	5 4) 🛱 🙀 📘	e

Figure III.49: Velocity magnitude variable chosen to color the fluid domain.

Contour plots of the velocity magnitude are then displayed in the **ParaView scene viewer**. By default, the colour scale is set to the **Cool to Warm** (RGB) colour scheme which is inadequate.

Coloring				
礡 Velocity	~	Magnitude	~	
🔒 Edit			-	
Styling		Choose pro	eset	

Figure III.50: Color map editor.

To modify the visualisation colour scheme, click on the **Choose Preset** button (see Figure III.50) button and select **Blue to Red Rainbow** (HSV) in the list of **Preset Color Scales**. Press **Apply** and then **Close**.





Figure III.51: Selecting the visualisation colour defaults.

Press the **Save current display settings values as default** button to save the changes. The contour plot of velocity magnitude is now updated for the new colour scale (Figure III.52).



Figure III.52: Contour plot of velocity magnitude.

The contour plot indicates that there is a zone of higher velocity flow defined by the green and red zones surrounded by lower and no-velocity regions in blue. Consistent with the chosen boundary conditions, the maximum velocity is equal to 1.0 m/s at the top, sliding wall and decreases to zero at the other, non-slip walls.

With the HSV colour scale, the contour plot is now clear, with the different levels clearly differentiable, but the image looks tessellated. As *Code_Saturne* outputs data at cell centres in the results file, in ParaView each mesh cell is painted with a pixel of colour corresponding to the exact value of the variable in the cell. Whilst this cell-data visualisation mode is correct to examine exact values at cell centres, it can yield ragged images unrepresentative of the solution's higher-order spatial accuracy and cannot be used in ParaView to generate vectors and streamline plots. Instead, to produce smoother

images, and vector and streamline plots, the cell-data can be interpolated to cell vertices (corners) using the ParaView CellDatatoPointData filter.

Having clicked on **ExtractBlock1** in the Pipeline Browser to select it, in the top menu bar click on Filters Alphabetical CellDatatoPointData. The new object, which now stores the **RESULTS.case** data interpolated to cell vertices, is added in the Pipeline Browser (Figure III.53).

۲	🗄 💼 RESULTS.case
۲	n ExtractBlock1
۲	i CellDatatoPointData1

Figure III.53: CellDatatoPointData filter added to the Pipeline Browser.

With **CellDatatoPointData** selected in the Pipeline Browser, click **Apply** in the **Properties** tab of the **Object Inspector** to bring up the smoothed-out contour plot in the **ParaView scene viewer**.



Figure III.54: ParaView short-cuts.

Picking the different variables to map, either from the **Display** tab in the Object Inspector for the **CellDatatoPointData** object or directly from the drop-down list in the menu bar for contour plots at the top of the SALOME window (see Figure III.54), create contour plots for the X and Y velocity components, as shown in Figure III.55 below. The plots use a **Surface** mode of **Representation**, which can be selected either from the **Display** tab and **Style** category, or directly from the drop-down list in the contour plot menu bar (see Figure III.54).



Figure III.55: Contour plots of the X (left) and Y (right) velocity components.

As expected, the maximum velocity in the X direction occurs at the top wall, where it nears 1.0 m/s (Figure III.55 (left)). The locations of the blue (flow in the negative direction) and red (flow in the positive direction) areas in both plots (Figure III.55) indicate that the flow is entrained by the top wall in a clockwise circular motion inside the box. These plots are useful to verify overall velocity magnitudes and patterns but, to get a better visualisation of the flow, one can create 2D glyphs which shows velocity vectors and streamlines which will show fluid particles trajectories.

To create the 2D glyphs, select CellDatatoPointData in the Pipeline Browser and click on Filters O Common Olyph in the top menu. The Glyph1 object is then added under CellDatatoPointData

- Properties (Glyph1)						
Glyph Source						
Glyph Type	2D Glyph	1				~
Glyph Type	Arrow				~	
□ Filled						
Center 0		0 0)	
Active Attributes						
Scalars		• Pres	sure			~
Vectors		 Velocity 			~	
Orientation						
🗹 Orient						
Scaling						
Scale Mode		vector 🗸				~
Scale Factor			0.1		x ~	C

Figure III.56: Glyph type and Scale Mode.

In the Object Inspector, select the **Display** tab. In the **Coloring** category, choose **Solid Color** and set this color to black by clicking on the **Edit color map** button (see Figure III.57).Press OK to finish.

	Basic colors
Coloring	Pick Screen Color
Solid Color	
Edit 🚔 🛱 🙀 📳 🖢	Hu <u>e</u> : 0 0 0 0
	Custom colors Sat: 0 0 Creen: 0 0
	Add to Custom Colors
	● Cancel

Figure III.57: Glyph coloring.

Conclude by pressing the Apply button. The result should look like Figure III.58.



Figure III.58: Contour plot of the velocity with 2D glyphs.

To make it possible to create and locate the streamlines with regard to the Cavity, you are going to create a combined image showing the streamlines superimposed on top of the mesh.

First, create the streamlines. With **CellDatatoPointData** selected in the Pipeline Browser, click on Filters Common Stream Tracer in the top menu. The **SteamTracer** object is then added under **CellDatatoPointData** in the Pipeline Browser. Keep the **CellDatatoPointData** object selected and visible (icon in bold next to the object). In the Object Inspector, select the **Display** tab. In the **Coloring** category, choose **Solid Color** in the **Color by** drop-down list. In the **Style** category, change the **Representation** to **Wireframe** and decrease the **Opacity** to 0.1 (Figure III.59).

🗖 Display (UnstructuredGrid					
Representation	Wireframe		~		
Coloring					
Solid Color	~		~		
Par Edit			le		
Styling					
Opacity	-0	0.1			
Line Width	1				

Figure III.59: Selecting the opacity.

The mesh lines displayed in the **ParaView scene Viewer** should be painted in faint black colour.

Next, select the **StreamTracer1** object in the Pipeline Browser. In the **Object Inspector**, select the **Properties** tab and modify the default settings for **Seeds**. Select **Point Source** for the **Seed Type**, and modify the X and Y coordinates of the seed point to 0.15 and 0.05, respectively. Request 60,000 points (**Number of Points**) and a radius of 1.0 (**Radius**), as shown in Figure III.60 below. Press **Apply** to validate your changes.

Seeds					
Seed Type Point Source					
Point Parameters					
Show Point					
Center 0.15 0.05 0.00499999					
Note: Use 'P' to pick 'Center' on mesh or 'Ctrl+P' to snap to the closest mesh point					
Center on Bounds					
Number Of Points 6000	0				
Radius		x v C			

Figure III.60: Selecting the streamlines settings.

In the **Display** tab, choose **Velocity** and **Magnitude** from the **Color** by drop-down lists of category **Color** and visualise the streamlines coloured by velocity magnitude, on top of the computational mesh (Figure III.61).



Figure III.61: Streamlines coloured by velocity magnitude superimposed on the mesh.

In addition to the general circular motion which had been inferred from the contour plots of velocity components, the streamlines now reveal two recirculation pockets in the bottom corners of the Cavity.

To further compare the calculated data with the available data from [4], create line plots of the X and Y velocities along the Y-axis and X-axis profiles which were set in the *Code_Saturne* GUI. Using the menu File Open ParaView File, import both the XVel_YaxisCentreLine.csv and the YVel_XaxisCentreLine.csv files. To read in the data and ensure that the velocities are correctly interpreted as real numbers by ParaView, in the Properties tab, define the Field Delimiter Characters as ',': a blank space followed by a comma. Also, select Merge Consecutive Delimiters (Figure III.62).

- XVel_YaxisCenterLine.csv
Vel_XaxisCenterLine.csv
Properties @ 🗵
Apply @Reset Relete ?
Search (use Esc to clear text)
= Properties (YVel_XaxisCente
✓ Detect Numeric Columns
✓ Use String Delimiter
✓ Have Headers
Field Delimiter , Characters
✓ Merge Consecutive Delimiters

Figure III.62: Specifying the velocity profiles.

Go to the **Information** tab and verify that all the data arrays are read in as **double**. The steps to create and label the graphs are similar to those already detailed for the graphs of the **monitoring points** data and they are not repeated here.

For direct comparisons with the plots from [4], for the graphs of X velocity, plot the Y coordinate as a function of Velocity[0]. For the graphs of Y velocity, plot Velocity[1] as a function of the X coordinate. The reference data provided in [4] for the fine mesh results (129x129) is listed in Section 5. The data can be copied to a .csv file which is then imported in ParaView at the same time as the XVel_YaxisCentreLine.csv and the Y_VelXaxisCentreLine.csv files. Set the plots so that the Code_Saturne and the reference results are visualised in the same view, and display the reference points with a marker rather than a line. Each data set can then be compared directly with the reference results (Figure III.63).



Figure III.63: Comparison between *Code_Saturne* (solid lines) and reference [4] results (circles). Y coordinate versus X velocity (left) and Y velocity versus X coordinate (right).

Overall, good agreement is obtained with the reference results [4], even though the results with *Code_Saturne* were obtained on a coarser mesh. Running on a finer mesh would make it possible to capture the velocity extremas with increased accuracy.

4 References

- [1] www.salome-platform.org
- F. ARCHAMBEAU, N. MÉCHITOUA, M. SAKIZ, Code_Saturne: a Finite Volume Code for the Computation of Turbulent Incompressible Flows -Industrial Applications, International Journal on Finite Volumes, Vol. 1, 2004.
- [3] www.code-saturne.org
- [4] U. GHIA, K.N. GHIA, and C.T. SHIN, High-Re solutions for incompressible flow using the Navier-Stokes equations and a multigrid method, Lowmal of Comp. Phys. Vol. 48, pp. 287–411, 1082.

Journal of Comp. Phys., Vol. 48, pp. 387-411, 1982.

 U. AYACHIT, *The ParaView Guide: A Parallel Visualization Application*, Kitware, 2015, ISBN 978-1930934306

5 Reference Data from [4]

To make it easier to replicate the comparative plots of the *Code_Saturne* and reference results, the data from [4] (Tables I and II) at Re = 1000 is reproduced in Figure III.3 below. The reference results listed in [4] were obtained on a fine, 129x129 mesh.

Y (m)	$\mathbf{U_x}~(\mathbf{m/s})$	X (m)	$\mathbf{U_y}~(\mathbf{m/s})$
1.00000	1.00000	1.00000	0.00000
0.97660	0.65928	0.96880	-0.21388
0.96880	0.57492	0.96090	-0.27669
0.96090	0.51117	0.95310	-0.33714
0.95310	0.46604	0.94530	-0.39188
0.85160	0.33304	0.90630	-0.51550
0.73440	0.18719	0.85940	-0.42665
0.61720	0.05702	0.80470	-0.31966
0.50000	-0.06080	0.50000	0.02526
0.45310	-0.10648	0.23440	0.32235
0.28130	-0.27805	0.22660	0.33075
0.17190	-0.38289	0.15630	0.37095
0.10160	-0.29730	0.09380	0.32627
0.07030	-0.22220	0.07810	0.30353
0.06250	-0.20196	0.07030	0.29012
0.05470	-0.18109	0.06250	0.27485
0.00000	0.00000	0.00000	0.00000

Table III.3: X velocity versus Y at the vertical mid-plane and Y velocity versus X at the horizontal mid-plane. Data reproduced from [4]. Re = 1000. 129x129 mesh.